SYNTHESYS OF RADAR ABSORBING MATERIALS FOR STEALTH AIRCRAFT BY USING NANOMATERIALS AND EVOLUTIONARY COMPUTATION

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Abstract

The present research shows a powerful method for the design of multi-layered radar absorbing materials useful in low observable and stealth manned or unmanned aircraft systems. The imitation of a desired behavior of the reflection coefficient allows to modeling the optimal arrangement of each layer in terms of thickness. material and Asignificant enhancement of the electromagnetic absorption is obtained through the use of different kind of carbon nanomaterials uniformly dispersed within an epoxy matrix. Several carbon-based nanocomposite materials are dielectrically characterized in the frequency range 2-18 GHz, and an evolutionary algorithm is used to run the mathematical model in order to find out the optimal solution in terms of the multi-layer composition. Interesting applications of the proposed design method are suggested other types of military targets in the marine and terrestrial scenario where the reduction of the radar observability is mandatory.

1 General Introduction

In the last 60 years the evolution of radar absorbing materials (RAM) [1-4] allowed to reduce the radar detection of military targets. In particular, the geometrical shape and the materials adopted are the main actors in stealth fighters and low observable aircrafts. The goal is to reduce the radar cross section [4] of the

target. The frequency of the detecting radar is one of the main issues in RAM design. In fact, the lower the frequency, the higher are generally the RAM's thickness and mass needed to accomplish the electromagnetic absorption requirements. Moreover. parameters must be carefully taken into account because they are potentially in conflict with the aerodynamics and payload capability of the aircraft. Radar systems work in a wide band of transmitted frequencies. The higher frequency of a radar system, the more it is affected by weather conditions such as rain or clouds. But the higher the transmitted frequency, the better is the accuracy of the radar system. At low frequency, radars are able to detect the target at great distances but with poor capability in recognizing the target details, i.e, topology and its potential the airplane harmfulness. At high frequencies radars are able to provide more details, but at short distances [4]. Nowadays, employed stealth technologies have not the desired effect at extremely low frequencies (below 300 MHz). The most used frequencies are in the bands above 2 GHz up to 18 GHz.

In stealth technology the requirement is a wideband radar invisibility. In low observability technology the condition is less stringent, since it is restricted to reduce the radar detection at short distances, mainly in the frequency range 8–12 GHz. In low observability technology the target is detected at long distance but with few details. When the target gets at shorter distance,

the radar is no more able to detect it since the low reflectivity determines a like noise-floor response. In turns, higher frequencies allow to design thinner and lighter RAM, which is essential for aircrafts. In Table 1 the NATO electronic warfare (EW) RF band designations are reminded, while in Table 2 the frequency bands and the radar operational propagation limitations [5] are reported.

Table 1: NATO Electronic Warfare (EW) RF Band Designations

Radar	ITU	IEEE	Wireless
Designation	Designation	Designation	Bands
HF	HF	A	
3-30MHz	3-30MHz	0-250MHz	
Not	VHF		
designated	30-300MHz		
P		В	
216-	UHF	250-500MHz	
450MHz	300-		
Not	3000MHz	С	
designated		500 – 1000MHz	802.15.4
L		D	
1-2GHz		1-2GHz	
S		E	802.11b,
3-4GHz		3-3GHz	802.11g,
			802.11n
			802.15.1,
			Bluetooth,
			802.15.4
	SHF	F	
	3-30GHz	3-4GHz	
С		G	
3-8GHz		3-6GHz	802.11a,
			802.11k
		Н	
		6-8GHz	
X		I	
8-12.4GHz		8-10GHz	
		J	
		10-20GHz	
J / Ku			
12.4 –			
18GHz			
K			
18-26.5GHz		K	
		20-40GHz	
Q / Ka			
26.5 -	EHF		
40GHz	30-300GHz		

In G- Band there are many mobile military battlefield surveillance, missile-control and ground surveillance radar sets with short or medium range. The size of the antennas provides for excellent accuracy and resolution. The influence of bad weather conditions is very high. Therefore, air-surveillance radars often use an antenna feed with circular polarization. This frequency band is predetermined for most types of weather radar used to locate precipitation in temperate zones like Europe.

In I/J (X- and Ku- Band radar) frequencyband (8 to 18 GHz) the relationship between wavelength and antenna size is considerably better than in lower frequency-bands. The I/J-Band is a relatively popular radar band for military applications like airborne radars for performing the roles of interceptor, fighter, and attack of enemy fighters and ground targets. A very small antenna size provides good performances. Missile guidance systems at I/J-Band are of convenient size and are, therefore, of interest for applications where mobility and light weight are important, and where a very long range is not a major requirement. This frequency band is widely used for maritime civil and military navigation radars. Very small and cheap antennas with high rotation speed are adequate for fair maximum ranges and good accuracy. Slotted waveguides and small patch antennas are used as radar antennas, mostly under protective radomes. This frequency band is also popular for spaceborne or airborne imaging radars based on synthetic aperture radar (SAR), both for military electronic intelligence and civil geographic mapping. A special inverse synthetic aperture radar (ISAR) is in use as a maritime airborne instrument for pollution control.

In the K- Band (K- and Ka- Band radar), the higher the frequency, the higher is the EM waves atmospheric absorption and attenuation. Otherwise, the achievable accuracy and the range resolution rise too. Radar applications in this frequency band provide short range, very high resolution and high data renewing rate.

Table 2: Frequency Bands and Radar Operational Propagation Limitations

LF	Allocations are provided in the frequency
30–300 kHz	range but no radar usage or applications
ME	have been identified.
MF	Used by continuous wave (CW) radar
300–3000 kHz	systems for accurate position location.
	Very high noise levels are characteristic
TIE	of this band.
HF 3–30 MHz	Refractive properties of the ionosphere
3-30 MITZ	make frequencies in this band attractive for long-range radar observations of areas
	such as over oceans at ranges of approximately 500–2000 nautical miles.
	Only a few radar applications occur in
	1 3 6
	limitations frequently outweigh its advantages: very large system antennas
	are needed, available bandwidths are narrow, the spectrum is extremely
	congested with other users, and the
	external noise (both natural noise and
	noise due to other transmitters) is high.
VHF	For reasons similar to those cited above,
30–300 MHz	this frequency band is not too popular for
50 500 WILL	radar. However, long-range surveillance
	radars for either aircraft or satellite
	detection can be built in the VHF band
	more economically than at higher
	frequencies. Radar operations at such
	frequencies are not affected by rain
	clutter, but auroras and meteors produce
	large echoes that can interfere with target
	detection. There have not been many
	applications of radar in this frequency
	range because its limitations frequently
	outweigh its advantages.
UHF	Larger antennas are required at the lower
300-3000 MHz	end than at the upper end of the UHF
	band. As compared to the above bands,
	obtaining larger bandwidths is less
	difficult, and external natural noise and
	weather effects are much less of a
	problem. At the lower end, long-range
	surveillance of aircraft, spacecraft, and
	ballistic missiles is particularly useful.
	The middle range of this band is used by
	airborne and spaceborne SAR's. The
	higher UHF end is well suited for short to medium-range surveillance radars. □
SHF	Smaller antennas are generally used in
3 GHz=30 GHz	
3 GHz-30 GHz	this band than in the above bands.
3 GHz–30 GHz	this band than in the above bands. Because of the effects of atmospheric
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These radar sets are called surface movement radar (SMR) or airport surface detection equipment (ASDE). Using of very short transmitting pulses (few nanoseconds) affords a range resolution so that outlines of the aircraft can be seen on the radars display [6].

In this research the attention has been focused on RAMs able to absorb the EM waves in extended or in well selected frequency ranges. In particular, the EM properties of layered structures made of carbon based nanocomposite materials have been studied in the frequency range 2-18 GHz. By taking into account the dielectric parameters of these materials, and by using an appropriate mathematical model, the EM reflection coefficient (RC) at the first air-RAM interface has been simulated.

The layered RAM optimization has been addressed by searching the most appropriate layering of nanocomposite materials able to fit the simulated RC to an a priori established reflection coefficient target (RCT) profile. The concept of frequency selective material (FSM) is introduced recalling that of frequency selective surface (FSS) [7-11]. The entire optimization procedure was managed by an evolutionary algorithm able to link mathematical model to a database αf nanocomposite materials arrayed in terms of their dielectric permittivity. The optimization procedure consists in the minimization of an objective function (OF) which essentially matches the simulated RC with a desired RCT. As far as the background is concerned, a huge amount of papers on RAM can be found in the literature [12-36]: some of the most recent works employ evolutionary algorithms, but none of them makes use of a synthesis method like that proposed in the present research. In section II the materials employed and their dielectric characterization are described. In section III the mathematical model and the optimization algorithm are analyzed. In section IV the results obtained by the simulations of several optimized multilayered structures are presented and discussed.

2 Experimental: nanocomposites characterization

The carbon-reinforced nanocomposite materials are briefly presented. The composite matrix employed is the Tencate EX-1545. It is a 2-part highly modified tough Cyanate Ester resin formulated to have extremely low viscosity at room temperature. The low viscosity allows resin transfer molding (RTM) processing with minimal heat to the resin, which extends resin pot life and mold filling time for difficult and/or large RTM structures. Typical applications of Tencate EX-1545 are

- aircraft structures
- radomes and antennae
- low observables
- missile structures
- high performance and high heat commercial applications

Industrial grade multi walled carbon nanotubes (MWCNT) were employed: they are the NanocylTM NC 7000 (average diameter around 9.5 nm, average length 1.5 µm, carbon purity 90%, metal oxide 10%, surface area 250-300m²/g) supplied by Nanocyl. As reported in the NC 7000 technical datasheet, the volume resistivity (ohm·cm) and the surface resistivity (ohm·sq) of NC 7000 MWCNT-reinforced polymeric composites decrease from 10¹⁵ and 10^{13} to 10^3 and 10^2 respectively when the inclusion weight percentage increases from 0% to 3%. In addition, three more typologies of micro/nanoparticles, carbon-based carbon nanofibers (CNF), graphite nanoplatelets (GNP) and polyaniline (PANI), have been investigated in terms of their capability to affect the EM properties of the polymeric matrix. The CNFs were purchased at Sigma Aldrich: they are cylindrical nanostructures with graphene layers arranged as stacked cones, cups or plates. Due to their high electrical conductivity and high aspect ratio, the CNFs can impart equivalent electrical conductivity to a composite at lower loadings than conventional conductive fillers do. The GNPs were provided by XG Sciences (xGNP C-750): they are unique nanoparticles consisting of short stacks of graphene sheets having a platelet shape, now

becoming increasingly available. The GNPs are very thin flat particles (1-20 nm in thickness) with large diameters (1-50 µm). PANI emeraldine base powder with conductivity value of 4-6 S/m and 2-3 µm of particle size was purchased at Sigma Aldrich. PANI is a conducting polymer of the semi-flexible rod polymer family. Some scanning electron microscope (SEM) micrographs of the materials employed are reported in Fig. 1 while in Table 3 their morphological properties are summarized.

Table 3. Main physical characteristics of the carbon-based powder employed.

Carbon Material (wt%)	Dimension	Specific Surface Area (m²/g)	Aspect Ratio (L/D)
MWCNT	av. diameter 9.5 nm	250 ÷ 300	~ 150
$(0.5 \div 2.0)$	av. length 1.5 μm		
CNF	inner diameter	~ 60	10 ÷ 100
$(0.5 \div 3.0)$	0.5-10nm		
	outer diameter		
	80-200nm		
	length 0.5-20µm		
GNP	diameter 1-50µm	750	-
$(0.5 \div 3.0)$	platelet thickness 1-		
	20nm		
PANI	particle size 2-3µm	-	-
$(0.5 \div 2.5)$			

These carbon-based powders have been mixed in several wt% to the hosting polymeric matrix. A full description of the carbon nanocomposites processing is provided by the authors in previous works [37-43]. The materials have been dielectrically characterized in terms of electric permittivity by means of a vector network analyzer (Agilent PNA-L N5235) and a coaxial airline [38]. The results of the measurement of the electric permittivity have design the multilavered used to electromagnetic wave absorbers. In Fig. 2 the real and imaginary part of the relative electric permittivity of MWCNT-reinforced composites are shown. A complete report of all the carbon nanocomposite materials realized has been recently published [43]. From the plot of Fig. 2, it can be observed that an increase of the nanomaterials weight concentration determines values of higher the relative electric permittivity.

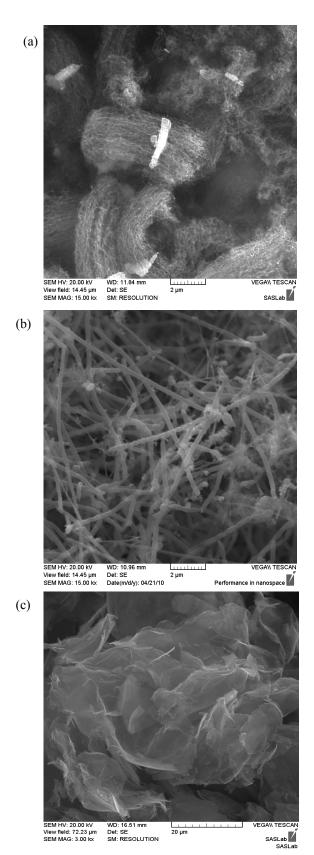
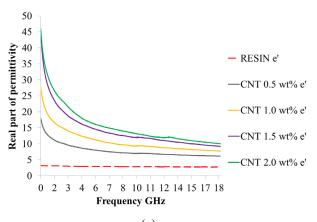


Fig. 1. SEM images of carbon nanostructures: (a) MWCNT, magnification 15 kx; (b) CNF, magnification 15 kx; (c) GNP, magnification 3 kx.



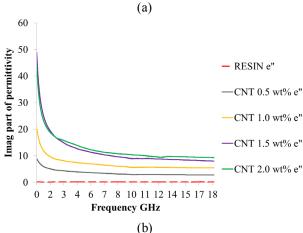


Fig. 2. Relative electric permittivity of MWCNT-reinforced composites: (a) real part, (b) imaginary part in the frequency range 300 MHz-18 GHz.

The possibility to join the most appropriate value of thickness with material type for each layer provides a full capability to design whatever kind of RAM. As one can guess, there are a lot of possible combinations: consequently, an upper limit has been imposed for the thickness of each layer and an iterative optimization approach has been used to the RAM design.

3 Design method

Each nanocomposite material has been coded to provide an univocal identification within the database:

- 1 → Resin Tencate EX-1545
- $2 \rightarrow MWCNT 0.5wt.\%$
- $3 \rightarrow MWCNT 1.0wt.\%$
- $4 \rightarrow MWCNT 1.5wt.\%$
- $5 \rightarrow MWCNT 2.0wt.\%$
- $6 \rightarrow \text{CNF } 0.5\text{wt.}\%$

 $7 \rightarrow$ CNF 1.0wt.% 8 → CNF 1.5wt.% $9 \rightarrow$ CNF 2.0wt.% $10 \rightarrow$ CNF 2.5wt.% $11 \rightarrow$ CNF 3.0wt.% $12 \rightarrow$ PANI 0.5wt.% $13 \rightarrow$ PANI 1.0wt.% $14 \rightarrow$ PANI 1.5wt.% $15 \rightarrow$ PANI 2.0wt% $16 \rightarrow$ PANI 2.5wt.% $17 \rightarrow$ GNP C750 0.5wt.% $18 \rightarrow$ GNP C750 1.0wt.% $19 \rightarrow$ GNP C750 1.5wt.% $20 \rightarrow$ GNP C750 2.0wt.% $21 \rightarrow$ GNP C750 2.5wt.% $22 \rightarrow$ GNP C750 3.0wt.%

A set of 22 different materials is thus available to build whatever FSM made of a certain number of layers, able to imitate a desired RCT profile. The search / evolutionary algorithm adopted to minimize the objective function must be able to easily go away from local minima. In this research, particle swarm optimization (PSO)[38],winning particle optimization (WPO)[42] and genetic algorithm (GA)[35,39] were used. Between these tools, as far as the authors experience is concerned, PSO seems to be able to find optimal solutions, minimizing the number of layers and the overall thickness of the FSM better than GA and WPO.

3.1 Mathematical model and layering optimization algorithm

The mathematical model of the multilayer structure has been already published [43]. It takes into account the dielectric parameters of each layer. The material of each layer is set by iterative optimization algorithm minimizing the final value of the objective function. The mathematical model and the iterative optimization algorithm can basically work in two main ways: one operates the RC minimization in a desired absolute frequency range [42], the other optimizes the RC by following a desired RCT profile [43]. In Fig. 3 two simple examples of expected results are depicted.

The expressions of the objective functions *OFs* able to perform the optimization

$$OF_1 = \min \left\{ \sum_{f} RC(f) \right\} \tag{1}$$

$$OF_2 = \min \left\{ \sum_{f} \frac{\left[RC(f) - RCT(f) \right]^2}{\left[RC(f) + RCT(f) \right]^2} \right\}$$
 (2)

(where f is the frequency step in the range considered) are extensively explained in [43].

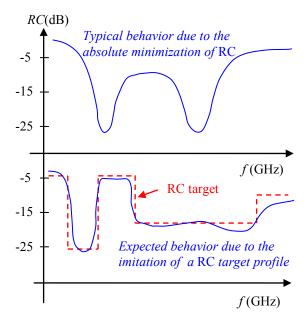


Fig. 3. Typical and expected behavior of the RC curves in the layered RAM optimization by using iterative optimization algorithms. Above, the RC curve is an example of typical result of the absolute minimization of the RC function. Below, the expected RC profile obtained by fitting a RCT is shown.

4 Results

4.1 RC absolute minimization (OF_1)

In this section some results obtained by using OF_1 are shown. Table 4 shows five RAM simulations. The growing thickness of RAM is controlled by a parameter within the OF_1 able to weight the overall thickness vs the EM absorption.

7D 11 4	D 4 3 4	1		\circ
Table 4.	KAM	simulations	using	OF1.

RAM	Materials Thickness Total				
Simulation	and layers	of layers	Thickness		
	una myors	(mm)	(mm)		
RAM n°1	MWCNT 2.0wt%	0.33	≅ 1.94		
(Fig. 4)	MWCNT 1.5wt%	0.17			
()	MWCNT 0.5wt%	1.43			
	(PEC)				
RAM n°2	CNF 3.0wt%	2.03	≅ 2.00		
(Fig. 5)	(PEC)				
RAM n°3	PANI 2.5wt%	3.0	≅ 5.15		
(Fig. 6)	MWCNT 2.0wt%	0.14			
	MWCNT 1.0wt%	2.01			
	(PEC)				
RAM n°4	Resin	2.93	≅ 8.36		
(Fig. 7)	MWCNT 0.5wt%	2.21			
	MWCNT 1.5wt%	0.65			
	MWCNT 2.0wt%	2.55			
	(PEC)	7.11			
RAM n°5	PANI 2.0wt%	3.5	≅ 14.9		
(Fig. 8)	MWCNT 0.5wt%	1.6			
	PANI 2.0wt%	0.7			
	MWCNT 1.0wt%	3.2			
	MWCNT 0.5wt%	3.9			
	MWCNT 2.0wt%	1.5			
	CNF 2.0wt%	0.1			
	CNF 3.0wt%	0.4			
	(PEC)				

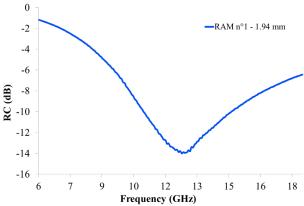


Fig. 4. RAM n°1 thick 1.94 mm, obtained with OF1. Frequency range 6-18 GHz.

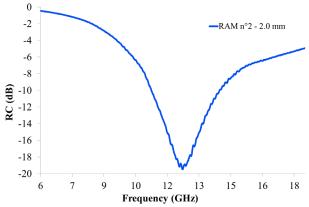


Fig. 5. RAM n°2 thick 2.0 mm, obtained with OF1. Frequency range 6-18 GHz.

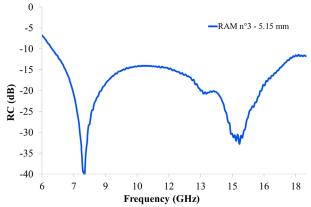


Fig. 6. RAM n°3 thick 5.15 mm, obtained with OF1. Frequency range 6-18 GHz.

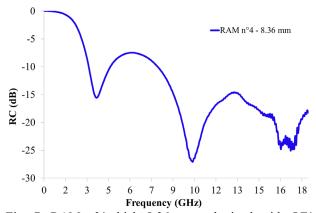


Fig. 7. RAM $n^{\circ}4$ thick 8.36 mm, obtained with OF1. Frequency range 300 MHz-18 GHz.

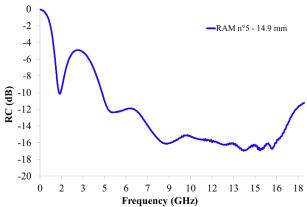


Fig. 8. RAM n°3 thick 14.9 mm, obtained with OF1. Frequency range 300 MHz-18 GHz.

By observing the results, it's clear that the minimization of OF_I does not allow to decide in advance the frequency where the RC will assume the relative minimum values. By increasing the thickness of RAM the absorption bandwidth increases too, but this method lacks EM absorption frequency optimization. The next paragraph deals with such issue.

4.2 Frequency selective optimization (OF_2)

The layering and the EM performance of five RAMs synthetized by applying OF₂ are shown. Table 4 reports materials and layers thickness for each RAM optimized as FSM (i.e., to follow a specific RCT).

Table 4	DAM	simulations		OE
1 able 4.	KAW	Simulations	using	$U\Gamma_2$.

RAM	Materials Thickness Total			
Simulation	and layers	of layers	Thickness	
Simulation	and layers	(mm)	(mm)	
RAM n°6	GNP 0.5wt%	3.80	≅ 14.9	
(Fig. 9)	MWCNT 1.5wt%	0.27	= 14.9	
(1 15. 7)	CNF 3.0wt%	1.09		
	GNP 2.0wt%	0.21		
	MWCNT 1.5wt%	5.99		
	PANI 2.0wt%	3.52		
	(PEC)	3.52		
RAM n°7	MWCNT 2.0wt%	0.20	≅ 2.9	
(Fig. 10)	MWCNT 0.5wt%	0.20		
()	Resin	1.10		
	MWCNT 2.0wt%	1.40		
	(PEC)			
RAM n°8	PANI 0.5wt%	3.43	≅ 28.8	
(Fig. 11)	CNF 2.5wt%	0.98		
	MWCNT 2.0wt%	1.00		
	PANI 2.5wt%	0.42		
	MWCNT 2.0wt%	0.83		
	PANI 1.0wt%	8.00		
	MWCNT 2.0wt%	1.16		
	CNF 1.0wt%	1.36		
	GNP 0.5wt%	5.10		
	CNF 2.0wt%	6.53		
	(PEC)			
RAM n°9	GNP 0.5wt%	3.25	≅ 20.6	
(Fig. 12)	CNF 1.0wt%	7.74		
	MWCNT 2.0wt%	3.32		
	CNF 3.0wt%	0.59		
	PANI 1.0wt%	5.69		
	(PEC)			
RAM nº10	CNF 1.5wt%	3.49	≅ 19.0	
(Fig. 13)	CNF 3.0wt%	1.68		
	CNF 2.5wt%	0.56		
	MWCNT 1.5wt%	9.00		
	PANI 0.5wt%	2.88		
	CNF 1.5wt%	1.44		
	(PEC)			

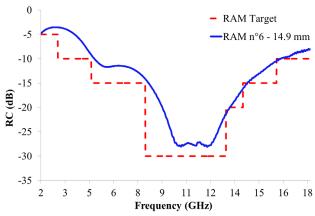


Fig. 9. RAM $n^{\circ}6$ thick 14.9 mm, obtained with OF_2 . Frequency range 2-18 GHz.

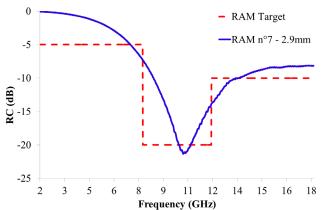


Fig. 10. RAM n°7 thick 2.9 mm, obtained with OF₂. Frequency range 2-18 GHz.

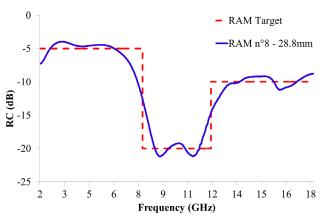


Fig. 11. RAM $n^{\circ}8$ thick 28.8 mm, obtained with OF₂. Frequency range 2-18 GHz.

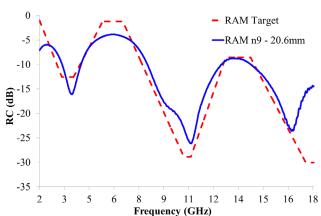


Fig. 12. RAM n°9 thick 20.6 mm, obtained with OF₂. Frequency range 2-18 GHz.

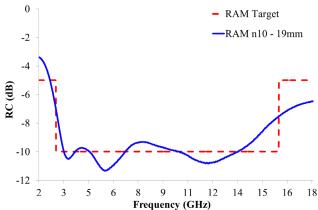


Fig. 13. RAM n°10 thick 19 mm, obtained with OF₂. Frequency range 2-18 GHz.

By observing the plots obtained minimizing OF_2 , it can be highlight the possibility to synthetize every kind of RC response. Such result is particularly important for applications of radar signature reduction in predefined frequency band, as reported in Tables 1,2.

The last result shown in Fig. 13 is quite interesting for extended absorption bandwidth applications. The thicknesses of the synthetized RAMs reported in Table 4 could appear inapplicable in some real scenarios where the maximum allowed thickness is of few mm. Nevertheless, it must be taken into account that by adding other kind of nanomaterials to the the EMlosses database. ofbasic nanocomposites could be enhanced. allowing to reduce the thickness of the single layers and of the final RAM.

5 Conclusion

In this work a methodology for multilayered RAM design by using the "synthesis" approach is described. This method need to access to an existing database of materials. Each material can be properly selected for a RAM layer. The materials in the database finally determine the RAM thickness and set the constraints which allow the synthesis to be successful in imitate a desired RC profile. The obtained results suggest this method as effective route to design advanced RAMs for military applications where FSMs seem to be the key to optimize the radar signature suppression performances. The

present research shows a powerful method for the design of multi-layered radar absorbing materials useful in low observable and stealth manned or unmanned aircraft systems.

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